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Parametric Design Optimization of Uncertain Ordinary Differential Equation Systems

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Abstract

This work presents a novel optimal design framework that treats uncertain dynamical systems described by ordinary differential equations. Uncertainty in multibody dynamical systems comes from various sources, such as: system parameters, initial conditions, sensor and actuator noise, and external forcing. The inclusion of uncertainty in design is of paramount practical importance because all real-life systems are affected by it. Designs that ignore uncertainty often lead to poor robustness and suboptimal performance. In this work uncertainties are modeled using Generalized Polynomial Chaos and are solved quantitatively using a least-square collocation method. The uncertainty statistics are explicitly included in the optimization process. Systems that are nonlinear, have active constraints, or opposing design objectives are shown to benefit from the new framework. Specifically, using a constraint-based multi-objective formulation, the direct treatment of uncertainties during the optimization process is shown to shift, or off-set, the resulting Pareto optimal trade-off curve.

A nonlinear vehicle suspension design problem, subject to parametric uncertainty, illustrates the capability of the new framework to produce an optimal design that accounts for the entire family of systems within the associated probability space.

Keywords: Design Optimization, Dynamic Optimization, Nonlinear Programming, Multi-Objective Optimization, Multibody Dynamics, Uncertainty Quantification, Generalized Polynomial Chaos

List of Variables (Nomenclature)

	variables
t	Time
ω	Random event
General	
x, X	Non-bolded variables generally indicate a scalar quantity
<i>x</i> , <i>X</i>	Bolded lower case variables are vectors, upper case variables are matrices
ξ	Random variable
xi	Bottom right index generally indicates a state (with occasional exceptions).
x ^j	Top right index generally indicates a stochastic coefficient, or mode.
. Υ	Bottom left index generally associates x to a specific collocation point
$a_{\gamma}^{k^{\gamma}}u_{\gamma}$	Ton left annotations indicate if a given variable is actuated or unactuated
χ, χ	The four maior maior is a stations
$\frac{\tilde{k}\chi_i}{\tilde{k}}$	
()'	Iranspose
$()_{\mathbf{q}}, \frac{\partial}{\partial a}$	Partial derivative notations
$()^{-1} ()^{\#}$	Matrix inverse and pseudo inverse
$r \overline{r}$	Lower and upper bounds on x
$\frac{\chi}{F[x]}$	Expected value or mean of r
$Var[x], \mu_{\chi}$	Variance of r
$ru[x], 0_{\chi}$	Standard Deviation of r
$sin[x], 0_X$ inf(x) sun(x)	Infimum and supremum of x
$\frac{\ln f(x)}{\ln d \cos \theta} d d$	
	Number of degrees of freedom (DOE)
$n_d \in \mathbb{N}$	Number of generalized coordinates, where $n > n$, dependent on kinematic representation of rotation
$n_{gc} \in \mathbb{N}$	Number of generalized coordinates, where $n_{gc} \ge n_d$, dependent on kinematic representation of rotation.
$n_s \in \mathbb{N}$	Number of states
$n_p \in \mathbb{N}$	Number of parameters $- c m^{n_i}$
$n_i \in \mathbb{N}$	Number of autoute $\mathbf{u} \in \mathbb{R}^n$
$n_0 \in \mathbb{N}$	Palmemial order
$p_0 \in \mathbb{N}$	Polynolillal oldel
$n_b \in \mathbb{N}$	Number of aellocation points
Dynamics	
$a \in \mathbb{D}^{n_{ac}}$	Independent generalized coordinates
$q \in \mathbb{R}^{n_{gc}}$	Independent generalized coordinates
$q \in \mathbb{R}^{n_{gc}}$ \dot{q}, \ddot{q} $n \in \mathbb{R}^{n_d}$	Independent generalized coordinates Rates and accelerations of generalized coordinates Constrained valorities
$ \begin{array}{c} \boldsymbol{q} \in \mathbb{R}^{n_{gc}} \\ \dot{\boldsymbol{q}}, \ddot{\boldsymbol{q}} \\ \boldsymbol{v} \in \mathbb{R}^{n_d} \\ \dot{\boldsymbol{v}} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Concerninged secondary
$q \in \mathbb{R}^{n_{gc}}$ q, \ddot{q} $v \in \mathbb{R}^{n_d}$ \dot{v} $q(0) = q$	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Generalized accelerations
	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Generalized accelerations Initial conditions
$ \begin{array}{c} \boldsymbol{q} \in \mathbb{R}^{n_{gc}} \\ \boldsymbol{\dot{q}}, \boldsymbol{\ddot{q}} \\ \boldsymbol{\upsilon} \in \mathbb{R}^{n_d} \\ \boldsymbol{\dot{\upsilon}} \\ \boldsymbol{q}(\boldsymbol{0}) = \boldsymbol{q}_0 \\ \boldsymbol{\upsilon}(\boldsymbol{0}) = \boldsymbol{v}_0 \\ \boldsymbol{\upsilon}(\boldsymbol{0}) = \boldsymbol{v}_0 \\ \boldsymbol{H} \in \mathbb{R}^{n_s \times n_s} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Generalized accelerations Initial conditions Kinematic mapping matrix relating rates of generalized coordinates to generalized velocities
$ \begin{array}{c} \boldsymbol{q} \in \mathbb{R}^{n_{gc}} \\ \dot{\boldsymbol{q}}, \ddot{\boldsymbol{q}} \\ \boldsymbol{v} \in \mathbb{R}^{n_d} \\ \dot{\boldsymbol{v}} \\ \boldsymbol{q}(\boldsymbol{0}) = \boldsymbol{q}_0 \\ \boldsymbol{v}(\boldsymbol{0}) = \boldsymbol{v}_0 \\ \boldsymbol{H} \in \mathbb{R}^{n_s \times n_s} \\ \boldsymbol{\theta} \in \mathbb{R}^{n_p} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Generalized accelerations Initial conditions Kinematic mapping matrix relating rates of generalized coordinates to generalized velocities Uncertain parameters
$ \begin{array}{c} q \in \mathbb{R}^{n_{gc}} \\ \dot{q}, \ddot{q} \\ \nu \in \mathbb{R}^{n_d} \\ \dot{v} \\ q(0) = q_0 , \\ v(0) = v_0 \\ H \in \mathbb{R}^{n_s \times n_s} \\ \theta \in \mathbb{R}^{n_p} \\ \tau \in \mathbb{R}^{n_i} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Generalized accelerations Initial conditions Kinematic mapping matrix relating rates of generalized coordinates to generalized velocities Uncertain parameters Input wrenches
$ \begin{array}{c} q \in \mathbb{R}^{n_{gc}} \\ \dot{q}, \ddot{q} \\ \upsilon \in \mathbb{R}^{n_d} \\ \dot{v} \\ q(0) = q_0 , \\ v(0) = v_0 \\ H \in \mathbb{R}^{n_s \times n_s} \\ \theta \in \mathbb{R}^{n_p} \\ \tau \in \mathbb{R}^{n_i} \\ \overline{\mathbf{M}} \in \mathbb{R}^{n_s \times n_s} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Initial conditions Kinematic mapping matrix relating rates of generalized coordinates to generalized velocities Uncertain parameters Input wrenches Square inertia matrix
$ \begin{array}{c} \boldsymbol{q} \in \mathbb{R}^{n_{gc}} \\ \dot{\boldsymbol{q}}, \ddot{\boldsymbol{q}} \\ \boldsymbol{v} \in \mathbb{R}^{n_d} \\ \dot{\boldsymbol{v}} \\ \boldsymbol{q}(\boldsymbol{0}) = \boldsymbol{q}_0 \\ \boldsymbol{\eta}(\boldsymbol{0}) = \boldsymbol{v}_0 \\ \boldsymbol{H} \in \mathbb{R}^{n_s \times n_s} \\ \boldsymbol{\theta} \in \mathbb{R}^{n_p} \\ \boldsymbol{\tau} \in \mathbb{R}^{n_i} \\ \boldsymbol{M} \in \mathbb{R}^{n_s \times n_s} \\ \boldsymbol{C} \in \mathbb{R}^{n_s} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Initial conditions Kinematic mapping matrix relating rates of generalized coordinates to generalized velocities Uncertain parameters Input wrenches Square inertia matrix Centrifugal, gyroscopic and Coriolis terms
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$ \begin{array}{c} \boldsymbol{q} \in \mathbb{R}^{n_{gc}} \\ \boldsymbol{\dot{q}}, \boldsymbol{\ddot{q}} \\ \boldsymbol{\dot{v}} \in \mathbb{R}^{n_d} \\ \boldsymbol{\dot{v}} \\ \boldsymbol{q}(\boldsymbol{0}) = \boldsymbol{q}_0 \\ \boldsymbol{\eta}(\boldsymbol{0}) = \boldsymbol{v}_0 \\ \boldsymbol{H} \in \mathbb{R}^{n_s \times n_s} \\ \boldsymbol{\theta} \in \mathbb{R}^{n_p} \\ \boldsymbol{\tau} \in \mathbb{R}^{n_i} \\ \boldsymbol{M} \in \mathbb{R}^{n_s \times n_s} \\ \boldsymbol{C} \in \mathbb{R}^{n_s} \\ \boldsymbol{N} \in \mathbb{R}^{n_s} \\ \boldsymbol{F} \end{array} $	Independent generalized coordinates Rates and accelerations of generalized coordinates Generalized velocities Generalized accelerations Initial conditions Kinematic mapping matrix relating rates of generalized coordinates to generalized velocities Uncertain parameters Input wrenches Square inertia matrix Centrifugal, gyroscopic and Coriolis terms Generalized gravitational and joint forces Differential operator
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Wi	Scalarlization weights for the individual input wrench contributions
С	Inequality constraints (typically bounding constraints)
a _i	Standard deviation scaling parameters

1 INTRODUCTION

1.1 MOTIVATION

Design engineers cannot quantify exactly every aspect of a given system. These uncertainties frequently create difficulties in accomplishing design goals and can lead to poor robustness and suboptimal performance. Tools that facilitate the analysis and characterization of the effects of uncertainties enable designers to develop more robustly performing systems. The need to analyze the effects of uncertainty is particularly acute when designing dynamical systems. Ultimately, if a robust system design is to be achieved, uncertainties must be accounted for up-front during the design process.

This work presents a novel parametric optimal design framework that treats uncertain dynamical systems described by linear or nonlinear ordinary differential equations (ODEs). System uncertainties, such as parameters, initial conditions, sensor/actuator noise, or forcing functions, are modeled using Generalized Polynomial Chaos (gPC) and are solved quantitatively using a least-square collocation method (LSCM). The computational efficiencies gained by gPC and LSCM enable the inclusion of uncertainty statistics in the optimization process.

Specifically, this work presents the new framework in both a nonlinear programming (NLP) and Directed Search (DS) optimization setting. The authors have found that the benefits of treating uncertainty during the parametric design optimization process are most evident when active constraints are present; therefore, particular attention is given to its use in a constraint-based formulation of *multi-objective optimization* (cMOO). These benefits are illustrated in an optimal vehicle suspension design case-study where the opposing performance criteria related to passenger ride comfort, suspension operating displacement, and road holding are simultaneously accounted for.

The author's prior work related to the motion planning of uncertain dynamical systems for fully-actuated and under-actuated systems was presented in [1-4].

1.2 STATE OF THE ART IN DYNAMIC DESIGN OPTIMIZATION AND UNCERTAINTY QUANTIFICATION

In the following, a review of the literature is presented where works related to dynamic design optimization and uncertainty quantification are specifically covered.

1.2.1 Dynamic Design Optimization

Many different formulations and solver techniques are presented in the literature for approaching system design through dynamic optimization. The selected case-study to showcase the new design framework of this paper is related to the optimal design of vehicle suspension systems; therefore, particular preference is given to the review of related literature.

A common theme is found in most of the vehicle suspension optimal design related works, namely, proper design of a suspension needs to address opposing design requirements related to passenger comfort (*ride*), suspension travel (*rattle*), and tire/road holding forces (*holding*). This necessitates a multi-objective approach to the optimal design problem. For example, [5-11] all employed linear vehicle models to construct a Pareto trade-off curve between the three referenced objectives.

Various methods were used to model the road input. A number of authors used stationary ergodic Gaussian inputs for linear models and used a power-spectral density (PSD) transformation of the system's linear frequency response [5, 6, 8, 11]. Additional attention was given to frequency weighted power-spectrum inputs based on standards such as ISO 2631 [6, 8, 11, 12]. This approach directly accounts for uncertainty in the road input of a linear system. Verros used the same Gaussian uncertain inputs for nonlinear quarter-car models through application of a Monte Carlo sampling technique [6]. These are examples of continuous road irregularity inputs. Additional authors treated isolated road irregularities such as speed bumps and potholes [12, 13].

Work related to active and semi-active suspension designs in addition to passive designs were presented in [7, 12-14]. The excellent work presented by Jazar [10] and by Gobbi and co-workers [5] approached the problem analytically; where a majority of the literature used numerical techniques showing a slight preference to DS-based (such as Genetic Algorithms [7, 9, 12]) versus NLP-based formulations [7, 13]. Also, adjacent applications related to rail [8, 15, 16] and bicycle [17, 18] optimal suspension designs were approached in a similar fashion.

1.2.2 Monte Carlo Uncertainty Quantification

The Monte Carlo (MC) method is considered the most robust method of uncertainty quantification. The method is quite simple; the probability space of the system is randomly sampled *n* times and statistical measures are determined from the ensemble [19]. MC provides a consistent error convergence rate independent of the number of uncertainties. However, the convergence rate of $1/\sqrt{n}$ is relatively slow.

Alternatively, quasi-Monte Carlo (QMC) methods deterministically sample the probability space with *low-discrepancy sequences* (LDS). QMC is reported to show improved constant convergence, $(\log n)^d/n$, for relatively low dimensional problems when compared to MC [20, 21]; where d is the number of dimensions.

1.2.3 Generalized Polynomial Chaos (gPC) Uncertainty Quantification

Generalized Polynomial Chaos (gPC) is a relatively new method that is rapidly being accepted in diverse applications. It's origins come from early work by Wiener in the the 1930's where he introduced the idea of *homogenous chaoses* [22]. His work made use of Gaussian distributions and the Hermite orthogonal polynomials. Xiu and Karniadakis generalized the concept by expanding the list of supported probability distributions and associated orthogonal polynomials [23, 24]; where the Galerkin Projection Method (GPM) was initially used. In [24-26], Xiu showed an initial collocation method based on Lagrange interpolation. A number of Collocation point selection methods were also show including tensor products and Smolyak sparse grids.

In [27], Sandu et. al. introduced the least-squares collocation method (LSCM) and used the roots of the associated orthogonal polynomials in selecting the sampling points. Cheng and Sandu showed the LSCM maintains the exponential convergence of GPM yet was superior in computational speed in [28]; where the Hammersley LDS data set was the preferred method in selecting collocation points. Cheng and Sandu also presented a modified time stepping mechanism where an approximate Jacobian was used when solving stiff systems.

1.2.4 Multi-Element gPC

The accuracy of gPC deteriorates over time in long simulations and is dependent on continuity of the system. In an effort to address these two concerns, Wan and Karniadakis developed multi-element gPC (MEgPC) [29, 30]. This method discretizes the probability space into non-overlapping partitions. Within each partition the traditional single element gPC is performed. Summing element integrations provides a complete integration of the full probability space. The algorithm presented adaptively partitioned the space based on estimates of error convergence. When an error estimate deteriorated to a specified point the element was split. The initial work was developed for the GPM methodology using uniform distributions. MEgPC was subsequently extended to arbitrary distributions in [31, 32]. Foo developed a collocation-based MEgPC in [33] and further extended the method to support higher dimensions using ANOVA methods in [34].

As an alternative to MEgPC, Witteveen and Iaccarino developed a similar multi-element method based on gPC called the simplex elements stochastic collocation (SESC) method. This method adaptively partitions the probability space using simplex elements coupled with Newton-Cotes quadrature. Their method has shown an O(n) convergence as long as the approximating polynomial order is increased with the number of uncertainties.

Another approach to addressing the long-term simulation accuracy of gPC was developed by Gerritsma and coworkers [35]; their approach constructs a new set of orthogonal polynomials adaptively as the probability density function (PDF) evolves with time. Their work has shown that the adaptively contrustructed polynomial basis allows for lower order approximations that have improved convergence for long-term simulations.

1.2.5 Recent Applications of gPC/MEgPC

The origins of gPC come from thermal/fluid applications; however, its adoption in other areas continues to expand. Sandu and coworkers introduced its application to multibody dynamical systems in [27, 28, 36-40]. Significant work has been done applying it as a foundational element in parameter [23-26, 41-59] and state estimation [60, 61], as well as system identification [62]. Relatively recent work has applied gPC to both classical and optimal control system design [41, 63, 64]. Also, MEgPC has been used applied to uncertainty quantification in power systems [65] and mobile robots [66].

The structure of this paper is as follows. A very brief introduction into the parametric optimal design of deterministic dynamical systems is reviewed in Section 2; Section 3 reviews the Generalized Polynomial Chaos methodology for uncertainty quantification; Section 4 introduces the new framework for parametric optimal design of uncertain ODE-based systems; and finally, Section 5 illustrates the strengths of the new framework through a cMOO vehicle suspension design case-study. Concluding remarks are found in Section 6.

2 PARAMETRIC OPTIMAL DESIGN OF DETERMINISTIC SYSTEMS

The new framework presented in this work is not dependent on a specific EOM formulation; formulations such as Newtonian, Lagrangian, Hamiltonian, and Geometric methodologies are all applicable. An Euler-Lagrange ODE formulation describes a multibody dynamical system [67, 68] by,

$$M(q(t),\theta(t))\dot{v}(t) + C(q(t),v(t),\theta(t))v(t) + N(q(t),v(t),\theta(t))$$

= $\mathcal{F}(q(t),v(t),\dot{v}(t),\theta(t)) = \tau(t)$ (1)

where $q(t) \in \mathbb{R}^{n_{gc}}$ are the generalized coordinates with $n_{gc} \ge n_d$; $v(t) \in \mathbb{R}^{n_d}$ are the generalized velocities and—using Newton's *dot* notation— $\dot{v}(t)$ contains their time derivatives; $\theta(t) \in \mathbb{R}^{n_p}$ includes system parameters of interest; $M(q(t), \theta(t)) \in \mathbb{R}^{n_d \times n_d}$ is the square inertia matrix; $C(q(t), v(t), \theta(t)) \in \mathbb{R}^{n_d \times n_d}$ includes centrifugal, gyroscopic, and Coriolis effects; $N(q(t), v(t), \theta(t)) \in \mathbb{R}^{n_d}$ are the generalized gravitational and joint forces; and $\tau(t) \in \mathbb{R}^{n_i}$ are the n_i applied wrenches (or torques). (For notational brevity, all future equations will drop the explicit time dependence.)

The relationship between the time derivatives of the independent generalized coordinates and the generalized velocities is,

$$\dot{\boldsymbol{q}} = \boldsymbol{H}(\boldsymbol{q}, \boldsymbol{\theta})\boldsymbol{v} \tag{2}$$

where $H(q, \theta)$ is a skew-symmetric matrix that is a function of the selected kinematic representation (e.g., Euler Angles, Tait-Bryan Angles, Axis-Angles, Euler Parameters, etc.) [3, 69, 70]. However, if (1) is formulated with independent generalized coordinates and the system has a fixed base, as in [1, 2], then (2) becomes $\dot{q} = v$.

The trajectory of the system is determined by solving (1)–(2) as an initial value problem, where $q(0) = q_0$ and $v(0) = v_0$. Also, the system measured outputs are defined by,

$$\mathbf{y} = \mathcal{O}(\mathbf{q}, \dot{\mathbf{q}}, \boldsymbol{\theta}) \tag{3}$$

where $\mathbf{y} \in \mathbb{R}^{n_o}$ with n_o equal to the number of outputs.

Given (1)–(3), the NLP-based formulation of the deterministic optimal design problem can be described as,

$$\begin{array}{ll}
\min_{x} & J = J_{obj} \\
\text{s.t.} & \mathcal{F}(q, v, \dot{v}, \theta) = \tau \\
& \dot{q} = H(q, \theta)v \\
& y = \mathcal{O}(q, \dot{q}, \theta) \\
& \mathcal{C}(y, \theta, \tau) \leq \mathbf{0} \\
& q(0) = q_{0}, q(t_{f}) = q_{t_{f}}
\end{array}$$
(4)

$$\dot{\boldsymbol{q}}(0) = \dot{\boldsymbol{q}}_{0}, \dot{\boldsymbol{q}}(t_{f}) = \dot{\boldsymbol{q}}_{t_{f}}$$

where $J = J_{obj}$ represents the design objective function; C is the list of design constraints; x is the list of optimization variables; and $\{q(0), \dot{q}(0), q(t_f), \dot{q}(t_f)\}$ are the system's initial conditions (ICs) and optional terminal conditions (TCs).

The NLP defined in (4) may be approached from either a sequential nonlinear programming (SeqNLP), or from a simultaneous nonlinear programming (SimNLP) perspective [71, 72]. (The literatures occasionally refers to the SeqNLP approach as partial discretization and to the SimNLP as full discretization [73].) In the SeqNLP approach, the dynamical equations (1)–(3) remain as continuous functions that may be integrated with standard off-the-shelf ODE solvers (such as Runge-Kutta). This yields a smaller optimization problem as only the optimization search/manipulated variables x are discretized. On the contrary, the SimNLP approach discretizes (1)–(3) over the trajectory of the system and treats the complete set of equations as equality constraints for the NLP. The discretized state variables are added to x to complete the full discretization. The SimNLP has a much larger set of constraints and optimization variables than the SeqNLP approach, but, enjoys a more structured NLP that typically experiences faster convergence.

The DS class of optimization solvers—techniques such as Genetic Algorithms, Differential Evolution, and Particle Swarm—typically only treat unconstrained optimization problems. Therefore, all the design constraints in (4) need to be converted from *hard* constraints to *soft* constraints; where *hard* constraints are explicitly defined as shown in (4), and constraints that are added to the definition of the objective function, J, are referred to as *soft* constraints. This is accomplished by additional cost terms of the form

$$J_{\text{constr}} = \sum_{i=1}^{n_c} \mu \max\left(0, \mathcal{C}_i(y, \theta, \tau)\right)^2 \tag{5}$$

where n_c represents the number of system constraints, and μ is a large constant. With a large μ , this relationship is analogous to an inequality like penalizing term, meaning, if the constraint C_i is outside its bounds—or outside the feasible region—then it's heavily penalized. When it's within the feasible region there is no penalty. Also, by squaring the *max* function its discontinuity is smoothed out; however, this is an optional feature and only necessary for a solver that uses gradient information.

Once the constraints have been converted to penalty terms, equation (4) can be reformulated as the following unconstrained optimization problem

$$\begin{aligned} \min_{x} & J = J_{obj} + J_{constr} \\ \text{s.t.} & \mathcal{F}(\boldsymbol{q}(t), \boldsymbol{v}(t), \dot{\boldsymbol{v}}(t), \boldsymbol{\theta}(t)) = \boldsymbol{\tau}(t) \\ & \dot{\boldsymbol{q}}(t) = \boldsymbol{H}(\boldsymbol{q}(t), \boldsymbol{\theta}(t)) \boldsymbol{v}(t) \\ & \boldsymbol{y}(t) = \mathcal{O}(\boldsymbol{q}(t), \dot{\boldsymbol{q}}(t), \boldsymbol{\theta}(t)) \end{aligned}$$

$$(6)$$

Equation (6) is analogous to the SeqNLP in that it is dependent on the ODE integration of the implicit dynamics found in (1)–(3) and their associated ICs.

Ultimately, the design task encoded in (4) and (6) is to determine what values of the manipulated variables x minimize J.

Design problems frequently have multiple objective terms defined in J_{obj}. Since the final cost function, J, ultimately needs to be a scalar value, a weighted scalarization of the various terms in J_{obj} is commonly used.

$$J_{obj} = w_1 J_1 + w_2 J_2 + \dots + w_n J_n$$
(7)

The relative weighting of the multiple terms, $\{w_1, w_2, ..., w_n\}$, yields a trade-off relationship between the various objective terms. In other words, there are an infinite set of optimal solutions—known as a Pareto optimal set—where each is uniquely defined by the scalarization weights. Problems of this nature are frequently referred to as *multi-objective optimization* (MOO) problems.

A Pareto optimal set may also be found as active constraint boundaries are moved. For example, if a given output, y_1 , is bounded from above by, \overline{y} , we have the following constraint, $y_1 \leq \overline{y}$. A Pareto set will be obtained from the optimal design for unique values of \overline{y} as long as the constraint is active. Once the constraint becomes inactive the constraint has no influence on the optimal value. Occasionally, a MOO problem may be appropriately rewritten such that extra objective terms, say $\{J_2, ..., J_n\}$, are redefined as problem constraints. For example,

$$\underbrace{J_2 - J_2 \le 0, \ J_2 - \bar{J}_2 \le 0}_{\vdots} \qquad (8)
 \underbrace{J_n - J_n \le 0, \ J_n - \bar{J}_n \le 0}$$

where $\{\underline{J}_i, \overline{J}_n\}$ represents lower and upper bounds. In doing so, the Pareto set is now governed by the unique values of the bounds of the only the active constraints. This reformulation of MOO will be referred to as a *constrained multi-objective optimization* (cMOO) problem. The cMOO reformulation will be of particular significance in Section 4 where the new framework for the optimal design of uncertain dynamical systems is presented.

The following section will briefly review Generalized Polynomial Chaos (gPC) which is the technique used for uncertainty quantification in the new framework presented in Section 4.

3 GENERALIZED POLYNOMIAL CHAOS

Generalized Polynomial Chaos (gPC), first introduced by Wiener [22], is an efficient method for analyzing the effects of uncertainties in second order random processes [23]. This is accomplished by approximating a source of uncertainty, θ , with an infinite series of weighted orthogonal polynomial bases called Polynomial Chaoses. Clearly, an infinite series is impractical; therefore, a truncated set of $p_o + 1$ terms is used with $p_o \in \mathbb{N}$ representing the *order* of the approximation. Or,

$$\theta(\xi) = \sum_{j=0}^{p_o} \theta^j \psi^j(\xi)$$
⁽⁹⁾

where $\theta^j \in \mathbb{R}$ represent known stochastic coefficients; $\psi^j \in \mathbb{R}$ represent individual single dimensional orthogonal basis terms (or modes); $\xi(\omega) \in \mathbb{R}$ is the associated random variable for θ that maps the random event $\omega \in \Omega$, from the sample space, Ω , to the domain of the orthogonal polynomial basis (e.g., $\xi: \Omega \to [-1,1]$).

Polynomial chaos basis functions are orthogonal with respect to the ensemble average inner product,

$$\langle \psi^{i}(\xi), \psi^{j}(\xi) \rangle = \int_{-1}^{1} \psi^{i}(\xi) \psi^{j}(\xi) w(\xi) d\xi = 0, \text{ for } i \neq j$$
(10)

where $w(\xi)$ is the weighting function that is equal to the joint probability density function of the random variable ξ . Also, $\langle \Psi^j, \Psi^j \rangle = 1, \forall j$ when using *normalized basis*; *standardized basis* are constant and may be computed off-line for efficiency using (10).

Generalized Polynomial Chaos can be applied to multibody dynamical systems described by differential equations [27, 36]. The presence of uncertainty in the system results in uncertain states. Therefore, the uncertain states can be approximated in a similar fashion as (9),

$$\dot{v}_i(\xi;t) = \sum_{j=0}^{n_b} \dot{v}_i^j(t) \Psi^j(\xi), \qquad i = 1 \dots n_s$$
(11)

where $\dot{v}_i^j(t) \in \mathbb{R}^{n_b}$ represent the gPC expansion coefficients for the i^{th} state; and $n_b \in \mathbb{N}$ represents the number of basis terms in the approximation. It is instructive to notice how time and randomness are decoupled within a single term after the gPC expansion. Only the expansion coefficients are dependent on time, and only the basis terms are dependent on the n_b random variables, ξ .

The stochastic basis may be multidimensional in the event that there are multiple sources of uncertainty. The multidimensional basis functions are represented by $\Psi^{j} \in \mathbb{R}^{n_{b}}$. Additionally, $\boldsymbol{\xi}$ becomes a vector of random variables, $\boldsymbol{\xi} = \{\xi_{1}, ..., \xi_{n_{p}}\} \in \mathbb{R}^{n_{p}}$, and maps the sample space, Ω , to an n_{p} dimensional cuboid, $\boldsymbol{\xi}: \Omega \to [-1,1]^{n_{p}}$ (as in the example of Jacobi chaoses).

The multidimensional basis is constructed from a product of the single dimensional basis in the following manner,

$$\Psi^{j} = \psi_{1}^{i_{1}} \psi_{2}^{i_{2}} \dots \psi_{n_{p}}^{i_{n_{p}}}, \quad i_{k} = 0 \dots p_{o}, k = 1 \dots n_{p}$$
⁽¹²⁾

where subscripts represent the uncertainty source and superscripts represent the associated basis term (or mode). A complete set of basis may be determined from a full tensor product of the single dimensional bases. This results in an excessive set of $(p_o + 1)^{n_p}$ basis terms. Fortunately, the multidimensional sample space can be spanned with a minimal set of $n_b = (n_p + p_o)!/(n_p! p_o!)$ basis terms. The minimal basis set can be determined by the products resulting from these index ranges,

$$\begin{split} i_1 &= 0 \dots p_o, \\ i_2 &= 0 \dots (p_o - i_1), \dots, \\ i_{n_p} &= 0 \dots \left(p_o - i_1 - i_2 - \dots - i_{(n_p - 1)} \right) \end{split}$$

The number of multidimensional terms, n_b , grows quickly with the number of uncertain parameters, n_p , and polynomial order, p_o . Sandu et. al. showed that gPC is most appropriate for modeling systems with a relatively low number of uncertainties [27, 36] but can handle large nonlinear uncertainty magnitudes.

Substituting (9) and (11) into (1) produces the following uncertain dynamics,

$$\mathcal{F}(\sum_{j=0}^{n_{b}} \mathbf{q}^{j}(t) \Psi^{j}(\xi), \sum_{j=0}^{n_{b}} \mathbf{v}^{j}(t) \Psi^{j}(\xi), \sum_{j=0}^{n_{b}} \dot{\mathbf{v}}^{j}(t) \Psi^{j}(\xi), \sum_{j=0}^{p_{o}} \boldsymbol{\theta}^{j}(t) \psi^{j}_{k}(\xi_{k})) = \boldsymbol{\tau}(t)$$
(13)

where the unknowns are now the unknown state gPC expansion coefficients.

The Galerkin Projection Method (GPM) is a commonly used method for solving (13); however, this is a very intrusive technique and requires a custom formulation of the dynamic EOMs. As an alternative, sample-based collocation techniques can be used without the need to modify the base EOMs.

Sandu et. al. [27, 28] showed that the collocation method solves formulations such as (13) by solving (1) at a set of points, $_k \mu \in \mathbb{R}^{n_p}$, $k = 1 \dots n_{cp}$, selected from the n_p dimensional domain of the random variables $\xi \in \mathbb{R}^{n_p}$. Meaning, at any given instance in time, the random variables' domain is sampled and solved n_{cp} times with $\xi = _k \mu$ (updating the approximations of all sources of uncertainty for each solve), then the uncertain coefficients can be determined at that given time instance. This can be accomplished by defining intermediate variables such as,

$$_{k}\dot{V}_{i}(t; _{k}\boldsymbol{\mu}) = \sum_{j=0}^{n_{b}} \dot{v}_{i}^{j}(t) \Psi^{j}(_{k}\boldsymbol{\mu})$$
⁽¹⁴⁾

where $i = 1 \dots n_s$, $k = 0 \dots n_{cp}$, and $l = 1 \dots n_l$. Substituting the intermediate variables into (13) yields,

$$\mathcal{F}\left({}_{k}Q_{i}(t; {}_{k}\boldsymbol{\mu}), {}_{k}V_{i}(t; {}_{k}\boldsymbol{\mu}), {}_{k}\dot{V}_{i}(t; {}_{k}\boldsymbol{\mu}), {}_{k}\Theta_{r}(t; {}_{k}\boldsymbol{\mu})\right) = \boldsymbol{\tau}(t)$$

$$i = 1 \dots n_{s}, k = 0 \dots n_{cp}, r = 1 \dots n_{p}$$
(15)

where $_kQ_i(t; _k\mu)$ and $_kV_i(t; _k\mu)$ are similar expansions as defined in (14). Also, each uncertain parameter is expanded with the single dimensional basis as,

$$_{k}\Theta_{r}(t; _{k}\boldsymbol{\mu}) = \sum_{j=0}^{p_{o}} \theta_{r}^{j}(t) \psi^{j}(_{k}\boldsymbol{\mu}).$$
⁽¹⁶⁾

Equation (15) provide a set of n_{cp} independent equations whose solutions determine the uncertain expansion coefficients. This is accomplished by recalling the relationship of the expansion coefficients to the solutions as in (14). In matrix notation (14) can be expressed for all states,

$$\dot{\boldsymbol{V}}_{i} = \left(\dot{\boldsymbol{v}}_{i}(t)\right)^{T} \boldsymbol{\Psi}(\boldsymbol{\mu}), \qquad i = 1 \dots {}^{u} \boldsymbol{n}_{s}$$
⁽¹⁷⁾

where the matrix,

$$A_{k,j} = \Psi^{j} \binom{k}{k} \mu, \ j = 0 \dots n_{b}, k = 0 \dots n_{cp}$$
⁽¹⁸⁾

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is defined as the *collocation matrix*. It's important to note that $n_b \le n_{cp}$. The expansion coefficients can now be solved for using (17),

$$u_i(t) = A^{\#} \dot{V}_i, \quad i = 1 \dots {}^u n_s$$
(19)

where $A^{\#}$ is the pseudo inverse of A if $n_b < n_{cp}$. If $n_b = n_{cp}$, then (19) is simply a linear solve. However, [28, 37-40] presented the least-squares collocation method (LSCM) where the stochastic state coefficients are solved for, in a least squares sense, using (19) when $n_b < n_{cp}$. Reference [28] also showed that as $n_{cp} \rightarrow \infty$ the LSCM approaches the GPM solution; by selecting $3n_b \le n_{cp} \le 4n_b$ the greatest convergence benefit is achieved with minimal computational cost. LSCM also enjoys the same exponential convergence rate as $p_o \rightarrow \infty$.

System outputs of (15) are analogous to (3); they are also uncertain and may equally be approximated in a similar fashion to (17) and subsequently solve for their expansion coefficients through (19).

The nonintrusive nature of the LSCM sampling approach is arguably its greatest benefit; (1) may be repeatedly solved without modification. Also, there are a number of methods for selecting the collocation points and the interested reader is recommended to consult [24-28] for more information.

4 OPTIMIZATION-BASED DESIGN OF UNCERTAIN DYNAMICAL SYSTEMS

The new framework for the optimal design of uncertain dynamical systems is now presented; this is a reformulation of (4) where IC, sensor, actuator, and parameter uncertainties are treated in a unified manner through the gPC techniques described in Section 3. The reformulation is,

$$\begin{array}{ll}
\min_{x} & \int_{obj}(t;\xi) \\
\text{s.t.} & \mathcal{F}(\boldsymbol{q}(t;\xi),\boldsymbol{v}(t;\xi),\dot{\boldsymbol{v}}(t;\xi),\boldsymbol{\theta}(t;\xi)) = \boldsymbol{\tau}(t;\xi) \\
& \dot{\boldsymbol{q}}(t;\xi) = H(\boldsymbol{q}(t;\xi),\dot{\boldsymbol{\theta}}(t;\xi))\boldsymbol{v}(t;\xi) \\
& \boldsymbol{y}(t;\xi) = \mathcal{O}(\boldsymbol{q}(t;\xi),\dot{\boldsymbol{q}}(t;\xi),\boldsymbol{\theta}(t;\xi)) \\
& \boldsymbol{c}(\boldsymbol{y}(t;\xi),\boldsymbol{\theta}(t;\xi),\boldsymbol{\tau}(t;\xi),t) \leq \mathbf{0} \\
& \boldsymbol{q}(0;\xi) = \boldsymbol{q}_{0},\boldsymbol{q}(t_{f};\xi) = \boldsymbol{q}_{t_{f}} \\
& \dot{\boldsymbol{q}}(0;\xi) = \dot{\boldsymbol{q}}_{0},\dot{\boldsymbol{q}}(t_{f};\xi) = \dot{\boldsymbol{q}}_{t_{f}}
\end{array}$$
(20)

The most interesting part of the new design framework comes in the ability to approach the design accounting for uncertainties by way of statistical moments of $\boldsymbol{\xi}$, such as expected values, variances, or standard deviations. These statistical moments may now be included in the definitions of the objective function, J(t; $\boldsymbol{\xi}$), and constraint equations, $\mathcal{C}(t; \boldsymbol{\xi})$. From [19], the statistical expected value is defined as,

$$\mu_x = E[x(\xi)] = \int_{\Omega} x(\xi) w(\xi) d\xi$$
(21)

and the variance,

$$\sigma_x^2 = Var[x(\xi)] = \int_{\Omega} (x(\xi) - \mu_x)^2 w(\xi) d\xi = E[(x(\xi) - \mu_x)^2]$$
(22)

with the standard deviation, $\sigma_x = \sqrt{Var[x(\xi)]}$. With these definitions new objective function terms may be defined. For example, the mean and standard deviation of an output may be efficiently computed by,

$$\mu_{\mathbf{y}} = E[\mathbf{y}(\mathsf{t};\boldsymbol{\xi})] = \mathbf{y}^{0}(\mathsf{t})\langle\Psi^{0},\Psi^{0}\rangle$$
(23)

$$\sigma_{\mathbf{y}} = \sqrt{E\left[\left(\mathbf{y}(\mathsf{t};\boldsymbol{\xi}) - \mu_{\mathbf{y}}\right)^{2}\right]} = \sqrt{\sum_{j=1}^{n_{b}} (\mathbf{y}^{j}(\mathsf{t}))^{2} \langle \Psi^{j}, \Psi^{j} \rangle}$$
(24)

Notice that due to the orthogonality of the polynomial basis these computations result in a reduced set of arithmetic operations on the respective expansion coefficients. Also, recall that $\langle \Psi^j, \Psi^j \rangle = 1, \forall j$ when using *normalized basis*; *standardized basis* are constant and may be computed offline for efficiency using (10). A number of efficient statistical quantities may be determined from the expansion coefficients. Additional examples of these statistical terms will be presented within the context of the case-study detailed in Section 5. Also, the authors presented a number of gPC based objective function and constraint equation terms in [1-4].

Equation (20) is the NLP formulation of the new framework for the optimal design of uncertain dynamical systems; it may also be solved through a SeqNLP or SimNLP approach as described in Section 2. The SeqNLP approach directly leverages the LSCM-based gPC solver, however, the SimNLP approach requires slight modification in the formulation to account for the full discretization of (1)–(3) in light of the LSCM technique.

$$\begin{aligned}
\min_{x} & \int_{obj}(t;\xi) \\
\text{s.t.} & \mathcal{F}\left(q(_{1}\mu), v(_{1}\mu), \dot{v}(_{1}\mu), \theta(_{1}\mu)\right) = \tau(_{1}\mu) \\
& \dot{q}(_{1}\mu) = H\left(q(_{1}\mu), \theta(_{1}\mu)\right) v(_{1}\mu) \\
& y(_{1}\mu) = \mathcal{O}\left(q(_{1}\mu), \dot{q}(_{1}\mu), \theta(_{1}\mu)\right) \\
& q(0;_{1}\mu) = q_{0,1}, \dot{q}(0;_{1}\mu) = \dot{q}_{0,1} \\
& \vdots \\
& \mathcal{F}\left(q\left(_{n_{cp}}\mu\right), v\left(_{n_{cp}}\mu\right), \dot{v}\left(_{n_{cp}}\mu\right), \theta\left(_{n_{cp}}\mu\right)\right) = \tau\left(_{n_{cp}}\mu\right) \\
& \dot{q}\left(_{n_{cp}}\mu\right) = H\left(q\left(_{n_{cp}}\mu\right), \theta\left(_{n_{cp}}\mu\right)\right) v\left(_{n_{cp}}\mu\right) \\
& y\left(_{n_{cp}}\mu\right) = \mathcal{O}\left(q\left(_{n_{cp}}\mu\right), \dot{q}\left(_{n_{cp}}\mu\right), \theta\left(_{n_{cp}}\mu\right)\right)
\end{aligned}$$
(25)

$$\boldsymbol{q}\left(0; _{n_{cp}}\boldsymbol{\mu}\right) = \boldsymbol{q}_{0,n_{cp}}, \boldsymbol{\dot{q}}\left(0; _{n_{cp}}\boldsymbol{\mu}\right) = \boldsymbol{\dot{q}}_{0,n_{cp}}$$

$$\mathcal{C}(\boldsymbol{y}(\boldsymbol{\xi}), \boldsymbol{\theta}(\boldsymbol{\xi}), \boldsymbol{\tau}(\boldsymbol{\xi}), t) \leq \boldsymbol{0}$$

Equation (25) duplicates the deterministic dynamical equations (1)–(3) n_{cp} times where each set has a unique collocation point, $_{k}\mu$. Each unique set of dynamical equations is then fully discretized and x is updated appropriately as described in Section 2. However, the system constraints, $\mathcal{C}(\mathbf{y}(\xi), \boldsymbol{\theta}(\xi), \boldsymbol{\tau}(\xi), t) \leq \mathbf{0}$, are calculated using the statistical properties determined by the LSCM and the n_{cp} sets of dynamical equations.

If a DS solver is to be used then (20) must be reformulated as an unconstrained problem. This requires all the hard design constraints in (20) to be converted to soft constraints by including them in the objective function. This may be accomplished in a similar fashion to the deterministic version presented in (5); the uncertain penalty-based constraints take the form of,

(1. 2)

$$const(t; \xi) = \sum_{i=1}^{n_c} \mu \max(0, C_i(t; \xi))^2$$
(26)

Once the constraints have been converted to penalty terms, equation (20) can be reformulated as an unconstrained optimization problem by,

I = I (4, 7) + I

$$\begin{array}{ll}
\min_{x} & J = J_{obj}(t;\xi) + J_{const}(t;\xi) \\
\text{s.t.} & \mathcal{F}(\boldsymbol{q}(t;\xi), \boldsymbol{v}(t;\xi), \dot{\boldsymbol{v}}(t;\xi), \boldsymbol{\theta}(t;\xi)) = \boldsymbol{\tau}(t;\xi) \\
& \dot{\boldsymbol{q}}(t;\xi) = H(\boldsymbol{q}(t;\xi), \boldsymbol{\theta}(t;\xi)) \boldsymbol{v}(t;\xi) \\
& y(t;\xi) = \mathcal{O}(\boldsymbol{q}(t;\xi), \dot{\boldsymbol{q}}(t;\xi), \boldsymbol{\theta}(t;\xi))
\end{array}$$
(27)

Again, equation (27) is analogous to the SeqNLP version defined in (20); it is dependent on the ODE integration of the implicit dynamics found in (1)–(3) and their associated ICs.

The new framework presented in (20), (25) or (27) and allows designers to directly treat the effects of modeled uncertainties during the optimal design process. The computational efficiencies of gPC enable the inclusion of statistical measures in objective function and constraint equations at a reduced computational cost as compared to contemporary techniques. However, the framework does introduce an additional layer of modeling and computation [2]. Therefore, it is of value to ask when the application of the new framework (20), (25), or (27) will yield a more robust design over the traditional deterministic optimal design approach presented in (4) and (6). Based on the authors' experience, the following general guidelines can help determine if a given design will benefit from the new framework:

- 1. System Nonlinearities: The probability density functions (PDFs) of uncertainties are likely to become skewed when propagated through a nonlinear system; the new framework will capture this information in the expected value measures.
- Active Constraints: Any design that has active constraints (i.e., $C_i(t; \xi) = 0$ for at least one *i*) will benefit from the inclusion of standard 2. deviation information in the constraint definitions; for example, $C_i(t; \xi) = (\mu_{y_i} \pm \sigma_{y_i}) - \overline{y}_i = 0$ will off-set the optimal design in a standard deviation sense to account for the entire family of realizable systems.
- Multi-objective Problems: Any multi-objective design that is reformulated as a cMOO will benefit from the added statistical information 3. of the new framework. This results in a shifted, or offset, Pareto optimal set (as will be illustrated in the case-study presented in Section 5). Application of the new framework directly to a penalty-based MOO formulated problem will likely not capture the Pareto offset determined by the cMOO formulation. This point is simply a restating of item #2 within the MOO context.

The new framework is a general formulation for the optimal design of dynamical systems described by ODEs. In an effort to show-case the benefits of the new framework a vehicle suspension optimal design problem is presented in Section 5.

5 AN ILLUSTRATIVE CASE-STUDY

This section presents an optimal design case-study of a passive nonlinear vehicle suspension as an illustration of the benefits that the general framework presented in Section 4 can provide. Many studies related to the optimal design of vehicle suspension parameters are found in the literature [6, 8, 9, 11-13]. Studies include linear and nonlinear vehicle models; MOO design; passive, semi-active, and active suspensions; and uncertain road inputs. This case-study is not comprehensive, but aims to illustrate the benefits of the new framework. As such, a nonlinear quarter-car suspension model was selected that is subject to parameter uncertainties. The literature frequently accounts for three conflicting objectives in a MOO design setting: the passenger comfort (ride); suspension displacement (rattle); and tire road holding forces (holding). These opposing objective terms yield the expected Pareto optimal set for a given parameter set; this case-study will address the optimal design through a cMOO.



Figure 1—An idealized 2-DOF deterministic quarter-car suspension model with a nonlinear asymmetric damper

5.1 VEHICLE SUSPENSION MODEL

An idealized two degree-of-freedom (DOF) nonlinear quarter-car suspension model shown in Figure 1 was used.

This system results in the following deterministic nonlinear dynamical equations-of-motion (EOMs),

$$\ddot{z}_{s} = -\frac{k_{s}}{m_{s}}(z_{s} - z_{u}) - \frac{1}{m_{s}}\mathcal{D}(\dot{z}_{s}, \dot{z}_{u})$$

$$\ddot{z}_{s} = \frac{k_{s}}{m_{u}}(z_{s} - z_{u}) + \frac{1}{m_{u}}\mathcal{D}(\dot{z}_{s}, \dot{z}_{u}) - \frac{k_{u}}{m_{u}}(z_{u} - z_{g})$$
(28)

The model has sprung and unsprung masses, m_s and m_u ; vertical mass positions about the equilibrium, $\{z_s, z_u\}$, and velocities $\{\dot{z}_s, \dot{z}_u\}$; suspension spring and damping coefficients, k_s and b_s ; tire spring coefficient, k_u ; and ground input position, z_g . The system is nonlinear due to the asymmetric damping force that is dependent on the velocity direction.

$$\mathcal{D}(\dot{z}_{s}, \dot{z}_{u}) = \begin{cases} b_{s}(\dot{z}_{s} - \dot{z}_{u}), & (\dot{z}_{s} - \dot{z}_{u}) \ge 0\\ \eta \ b_{s}(\dot{z}_{s} - \dot{z}_{u}), & (\dot{z}_{s} - \dot{z}_{u}) \ge 0 \end{cases}$$
(29)

The ratio of damping forces is determined by the scalar η .

The literature contains various methods for modeling the road input, z_g . A number of authors used stationary ergodic Gaussian inputs for linear quarter-car models through a power-spectral density (PSD) transformation of the system's linear frequency response [8, 11]. Additional attention was given to frequency weighted power-spectrum inputs based on standards such as ISO 2631 [6, 8, 11, 12]. This approach directly accounts for uncertainty in the road input of a linear system. Verros used the same Gaussian uncertain inputs for nonlinear quarter-car models through application of a Monte Carlo sampling technique [6]. These are examples of continuous road irregularity inputs. Additional authors treated isolated road irregularities such as speed bumps and potholes [12, 13]. These inputs were modeled by,

$$z_g = A\sin(\omega t) \tag{30}$$

where A represents the amplitude of the bump or pothole; $\omega = \frac{\pi v}{l}$ is the frequency of the irregularity determined by the vehicle velocity v and base length of the irregularity l; and t represents time.

This work chose to use a series of isolated road bumps defined by (30). Each bump is uniquely spaced with no overlap with one another and their amplitude was A = 0.15 meters. The frequencies of the speed bumps were selected to be $\omega = [1, 5, 10, 15]$ Hertz. Filtered Gaussian noise with a maximum amplitude of A = 0.03 meters was super-imposed over the series of speed bumps. The cut-off frequency of the filtered Gaussian noise was 35 Hertz. A representative road input signal is shown in Figure 2.

5.1.1 Optimal Design of Deterministic System

Contemporary optimal designs of a vehicle suspension commonly account for three opposing performance indexes: *ride*, *rattle*, and road *holding*. These performance indexes may be defined as,

$$J_{ride} = \int_0^{t_f} \ddot{z}_s^2 \, dt \tag{31}$$

$$J_{rattle} = \int_{0}^{t_{\rm f}} (z_s - z_u)^2 \, dt \tag{32}$$

$$J_{holding} = \int_0^{t_f} \left(z_g - z_u \right)^2 dt \tag{33}$$

The *ride* index aims to minimize the vertical accelerations experienced by a passenger. The *rattle* index aims to avoid the suspension displacement reaching its physical limits. The *holding* aims to minimize the variation of the dynamic force between the tire and the road [11]. Equation (33) is defined by the tire deflection. Given the linear relationship between the tire deflection and the tire/road force the same minimized variation is accomplished.





$$J = w_1 J_{ride} + w_2 J_{rattle} + w_3 J_{holding}$$
(34)

Such a definition results in the Pareto optimal set which depends on the weights $\{w_1, w_2, w_3\}$. As discussed in Section 2, the MOO can be reformulated in a constraint formulation yielding a similar Pareto optimal set. For example, J_{rattle} and $J_{holding}$ can be converted to *hard* constraints of the dynamic optimization problem. However, to reflect a more physical meaning of these qualities, (31)–(32) will be slightly redefined to root-mean-square (*rms*) values determined over the trajectory of the system.

$$J_{rattle} = rms(z_s - z_u) = \sqrt{\frac{1}{t_f} \int_0^{t_f} (z_s - z_u)^2 dt}$$
(35)

$$J_{holding} = rms(z_g - z_u) = \sqrt{\frac{1}{t_f} \int_0^{t_f} (z_g - z_u)^2 dt}$$
(36)

Hard constraints of the form presented in (8) and based on (35)-(36) can be defined as,

x

$$\underline{J}_{rattle} \le J_{rattle} \le J_{rattle} \tag{37}$$

$$J_{holding} - \bar{J}_{holding} \le 0 \tag{38}$$

where $\{\underline{J}_{rattle}, \overline{J}_{rattle}, \overline{J}_{holding}\}$ represent lower/upper bounding constraints. By sweeping through reasonable ranges for $\{\underline{J}_{rattle}, \overline{J}_{rattle}, \overline{J}_{holding}\}$ a Pareto optimal set may be found.

Therefore, the NLP-based cMOO formulation for the deterministic vehicle suspension design problem is,

r t_f

$$\begin{aligned} \min_{\substack{\{k_s,b_s\}}} & J = \int_0^{} \ddot{z}_s^2 dt = y_5 \\ \text{s.t.} & \ddot{z}_s = -\frac{k_s}{m_s} (z_s - z_u) - \frac{1}{m_s} \mathcal{D}(\dot{z}_s, \dot{z}_u) \\ & \ddot{z}_s = \frac{k_s}{m_u} (z_s - z_u) + \frac{1}{m_u} \mathcal{D}(\dot{z}_s, \dot{z}_u) - \frac{k_u}{m_u} (z_u - z_g) \\ & \mathbf{y} = [z_s, \dot{z}_s, z_u, \dot{z}_u, \int_0^{t_f} \ddot{z}_s^{-2} dt] \\ & J_{rattle} - \bar{J}_{rattle} \leq 0, \quad (extension) \\ & J_{rattle} - J_{rattle} \leq 0, \quad (compression) \\ & \bar{J}_{holding} - \bar{J}_{holding} \leq 0 \\ & \underline{k_s} \leq k_s \leq \bar{k_s} \\ & \underline{b_s} \leq b_s \leq \bar{b_s} \\ & \mathbf{z}(0) = z_0 \\ & \dot{\mathbf{z}}(0) = \dot{\mathbf{z}}_0 \end{aligned}$$

$$(39)$$

where $\{\overline{k}_s, \overline{k}_s\}$ and $\{\overline{b}_s, \overline{b}_s\}$ are reasonable physical bounds on the spring and damping coefficients, respectively; $\{z_0, \dot{z}_0\}$ are initial conditions for the vector of state variables, $\mathbf{z} = [z_s, \dot{z}_s, z_u, \dot{z}_u]^T$; and the list of solver manipulated variables is $\mathbf{x} = \{k_s, b_s\}$. Table 1 lists all nominal values for the system parameters and bounds for (39). Notice that the fifth output, $y_5 = \int_0^{t_f} \ddot{z}_s^2 dt$, is equal to the defined measure for the *ride* quality to be minimized.

Equation (39) may also be reformulated for non-gradient-based DS solvers as was presented in (6).

Table 1—System parameters, bounds, and uncertainties

Parameter	Mean (μ)	Std (σ)	Units (SI)
m_s	376	$m_{s}^{0}/8$	kg
m_u	$m_s/4$	-	kg
k _s	30,000	$k_{s}^{0}/8$	N/m
\overline{k}_s	300,000	-	N/m
\underline{k}_{s}	$\mathcal{F}(\overline{J}_{rattle}) = 23,339$	-	N/m
b _s	2,000	$b_{s}^{0}/8$	N-s/m
\overline{b}_s	50,000	-	N-s/m
\underline{b}_s	$1 + \sigma_{b_s}$	-	N-s/m
η	1.39	$\eta^0/8$	-
k _u	200,000	$k_{u}^{0}/8$	N/m
$\overline{J}_{holding}$	Figure 5	-	m
$J_{rattle}, \overline{J}_{rattle}$	Figure 4	-	m
$\{z_0, \dot{z}_0\}$	$zeros(n_p, 1)$	-	-
v	11.18	-	m/s

5.1.2 Optimal Design of Uncertain System

None of the cited works for optimal vehicle suspension design treated uncertain system parameters; however, the new framework presented in Section 4 is capable of treating system uncertainties originating from sensor outputs, actuator inputs, as well as system parameters and initial conditions within the unified gPC methodology (as presented in Section 3). Given the prior emphasis on input uncertainty in the literature, this work focuses on illustrating treatment of parametric uncertainty applied to a nonlinear system model. Varying passenger and cargo loads, fatiguing/deteriorating suspension components, and variations in tire air pressure are all very practical sources of uncertainty in a vehicle. Therefore, five system parameters were selected for this study, $\theta(\xi) = \{m_s(\xi), k_s(\xi), \eta(\xi), k_u(\xi), \}$. Each uncertain parameter is assumed to have a uniform distribution and is therefore modeled with a Legendre polynomial expansion. This takes the form of,

 θ_r

$$(\xi_r) = \theta_r^0 + \theta_r^1 \xi_r, \qquad r = 1 \dots n_p \tag{40}$$

Figure 3 illustrates the uncertain nonlinear model.



Figure 3—An uncertain 2-DOF quarter-car suspension model with a nonlinear asymmetric damper. The five uncertain parameters are, $\theta(\xi) = \{m_s(\xi), k_s(\xi), b_s(\xi), \eta(\xi), k_u(\xi)\}$.

This system results in the following set of uncertain nonlinear EOMs,

$$\ddot{z}_{s}(\xi) = -\frac{k_{s}(\xi)}{m_{s}(\xi)} \left(z_{s}(\xi) - z_{u}(\xi) \right) - \frac{1}{m_{s}(\xi)} \mathcal{D} \left(\dot{z}_{s}(\xi), \dot{z}_{u}(\xi) \right)$$

$$\ddot{z}_{u}(\xi) = \frac{k_{s}(\xi)}{m_{u}} \left(z_{s}(\xi) - z_{u}(\xi) \right) + \frac{1}{m_{u}} \mathcal{D} \left(\dot{z}_{s}(\xi), \dot{z}_{u}(\xi) \right) - \frac{k_{u}(\xi)}{m_{u}} \left(z_{u}(\xi) - z_{g}(\xi) \right)$$
(41)

and the corresponding NLP-based cMOO design problem is,

where the uncertain asymmetric damping force is,

$$\mathcal{D}(\dot{z}_{s}(\xi), \dot{z}_{u}(\xi)) = \begin{cases} b_{s}(\xi)(\dot{z}_{s}(\xi) - \dot{z}_{u}(\xi)), & (\dot{z}_{s}(\xi) - \dot{z}_{u}(\xi)) \ge 0\\ \eta(\xi) & b_{s}(\xi)(\dot{z}_{s}(\xi) - \dot{z}_{u}(\xi)), & (\dot{z}_{s}(\xi) - \dot{z}_{u}(\xi)) \ge 0 \end{cases}$$
(43)

The objective function is now a function of the uncertain *ride* comfort, (which is the fifth system output, $y_5(\xi) = \int_0^{t_f} (\ddot{z}_s(\xi))^2 dt$).

$$J(\xi) = E[y_5(\xi)] + \sqrt{Var[y_5(\xi)]} = \mu_{y_5} + \sigma_{y_5}$$
(44)

11

$$= y_5^0 \langle \Psi^0, \Psi^0 \rangle + \sqrt{\sum_{j=1}^{n_b} (y_5^j)^2 \langle \Psi^j, \Psi^j \rangle}$$

The rattle and holding constraints are also functions of the uncertainties. The uncertain rattle constraints may be defined as,

$$J_{rattle}(\xi) = \begin{cases} (\mu_{z_s} + a_1 \sigma_{z_s}) - (\mu_{z_u} - a_2 \sigma_{z_u}) - \bar{J}_{rattle} \le 0\\ \underline{J}_{rattle} - (\mu_{z_s} - a_3 \sigma_{z_s}) - (\mu_{z_u} + a_4 \sigma_{z_u}) \le 0 \end{cases}$$
(45)

where the various μ and σ computations take the form shown in (44) and defined in (23)–(24); the constants a_i represent scaling factors of the standard deviation. These constraints represent the extreme *rattle* conditions from a standard deviation perspective. The uncertain *holding* constraint is a function of the sixth system output, $y_6(\xi) = rms(z_u(\xi) - z_g)$ and may be defined as,

$$J_{holding}(\xi) = \left(\mathbb{E}[y_6(\xi)] + a_5 \sqrt{Var[y_6(\xi)]} \right) - \overline{J}_{holding} \le 0$$

$$= \mu_{y_6} + a_5 \sigma_{y_6} - \overline{J}_{holding} \le 0$$

$$= \left(y_6^0 \langle \Psi^0, \Psi^0 \rangle + a_5 \sqrt{\sum_{j=1}^{n_b} (y_6^j)^2 \langle \Psi^j, \Psi^j \rangle} \right) - \overline{J}_{holding} \le 0$$
(46)

Notice that due to the orthogonality of the polynomial basis the computations in (44)–(46) result in a reduced set of efficient operations on the respective expansion coefficients. No integrals are required and the statistical computations are relatively efficient.

It is important to re-emphasize that equations (45)–(46) constrain the system in a standard deviation sense. This means a subset of the systems within the probability space can still not satisfy the constraints. In order to guarantee that all systems within the probability space will satisfy the constraints, equations (45)–(46) would need to be redefined such that the *supremum* and/or *infimum* statistics are used instead of the standard deviations; however, the *supremum* and *infimum* are very expensive to calculate. The scaling constants, a_i , may be used to 'tune' the design such that a desired percentage of the systems from the probability space will satisfy the constraints. This point is illustrated in greater detail in the following Results Section.

Finally, the design objective of (42) is to determine mean values for the suspension components, $\mathbf{x} = \{k_s^0, b_s^0\}$, that minimize the *ride* being subject to the *rattle* and *holding* constraints; where $\{k_s^0, b_s^0\}$ are bounded by $\{\overline{k}_s, \overline{k}_s\}$ and $\{\overline{b}_s, \overline{b}_s\}$, respectively. Table 1 details all the uncertain parameters and associated bounds used in the case-study.

5.1.3 Results

Figure 4 and Figure 5 best illustrate the benefits of treating uncertainty during the optimal design process. These results clearly show that the presence of uncertainty in a system results in an off-set of the Pareto optimal design trade-off curve. Figure 4 is the 2D Pareto curve showing trade-off between the *ride* objective and the *rattle* constraint. Figure 5 shows a 2D Pareto curve perpendicular to that shown in Figure 4; this figure shows the trade-off between the *ride* objective and the *holding* constraint.

Both Figure 4 and Figure 5 show the Pareto curves flattening out at some point. This occurs when the system transitions from one active constraint to the other. Meaning, in Figure 4, the active constraint is the *rattle* constraint while the curve has a negative slope. However, when the slope flattens out the *holding* constraint is active. Since the *holding* constraint is constant in this plane the Pareto curve has a slope of zero. The same behavior is evident in Figure 5, however, the roles switch. First, the *holding* constraint is active and then the system transitions to the flat *rattle* constraint.



Rattle Constraint Bound (m)

Figure 4—A single 2D plane from the 3D Pareto optimal set showing the trade-off between the objective *ride* and *rattle* constraint; the *holding* constraint is held constant, $\overline{J}_{holding} = 0.034 [m]$. Both the deterministic (dOpt) and uncertain (uOpt) cases are shown. These results confirm that the presence of uncertainty requires an off-set of the Pareto optimal solution set to realize a more robust design. The set enclosed by the ellipse correspond to Figure 6.



Holding Constraint Bound (m)

Figure 5—A single 2D plane from the 3D Pareto optimal set showing the trade-off between the objective *ride* and *holding* constraint; both the compression and extension *rattle* constraints are held constant, $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.203 [m]$. Both the deterministic (dOpt) and uncertain (uOpt) cases are shown. These results confirm that the presence of uncertainty requires an off-set of the Pareto optimal solution set to realize a more robust design.

One distinct difference is apparent between Figure 4 and Figure 5; in Figure 4 the deterministic Pareto curve transitions its active constraints before the uncertain curve. However, Figure 5 has an opposite behavior; the uncertain curve transitions before the deterministic one. Figure 6 and Figure 7 help illustrate why this inconsistent behavior exists. Figure 6 presents a projection of the 3D optimal solution onto the three orthogonal 2D planes, {holding/rattle, holding/ride, rattle/ride}. When the bounding constraints are set to $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.203 \, [m]$ and $\overline{J}_{holding} = 0.034 \, [m]$ the deterministic optimal design has an active holding constraint where the uncertain optimal design has an active rattle constraint. Since the uncertain σ_{rattle}^* is so large, the optimal design was pushed to a significantly lower $\mu_{holding}^*$ value when compared to the deterministic $x_{holding}^*$ value; thus, the deterministic Pareto transitions first. This same behavior is apparent in Figure 7 which shows the 2D projection only in holding/rattle plane. Figure 5 presents the Pareto relating the holding constraint to the ride objective; so, as the holding bound increases Figure 7 shows how the uncertainty box enclosing the uncertain mean design—created by the standard deviation in the respective rattle and holding direction—encounters the rattle constraint before the corresponding deterministic solution does.



Figure 6—Projection of the 3D deterministic and uncertain solutions onto the three orthogonal 2D planes. This is an example of an optimal solution with an active *rattle* constraint. The constraint bounds are: $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.203 \ [m]$, and $\overline{J}_{holding} = 0.034 \ [m]$.



Figure 7—The 2D projection of Figure 5's 3D deterministic and uncertain solutions onto the *holding/rattle* plane. This shows the transition from an active *holding* constraint to an active *rattle* constraint as the *holding* bound is increased. The constraint bounds are: $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.203 \ [m]$, and $\overline{J}_{holding} = \{0.0305, 0.032, 0.0335, 0.035\} \ [m]$. The markers for the

deterministic and uncertain mean designs correspond within a given set. Also, the line of the uncertainty box associated with a given set matches line of the *holding* bound for that set.

The corresponding 2D trade-off curve in the parameter space for the case corresponding to Figure 5 with $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.203 \ [m]$ is shown in Figure 8. This figure is instructive in that it shows the optimal mean spring constant, $(k_s^0)^*$, is found to be at the lower bound, \underline{k}_s , for a majority of the designs. Only when $\overline{J}_{holding}$ is equal to the two lowest values shown in Figure 5 does the optimal mean spring constant leave the lower bounding constraint. This behavior makes sense in that the *ride* objective is most influenced by the k_s versus b_s [5]; therefore, as long as the active constraint can be satisfied with $k_s^0 = \underline{k}_s$ then the Pareto curve is largely defined by the damping mean, b_s^0 . Again, the design from the new framework shifts the optimal parameter set relative to that designed by a traditional deterministic optimal design.



Figure 8—Parameter Pareto trade-off curve when $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.203 \ [m]$.

The final point that illustrates the benefits of the new framework is shown in Figure 9; this figure shows the resulting *rattle* constraint values of the optimal deterministic design applied to an uncertain system. A 1000 sample Monte Carlo simulation shows that 59.6% of the systems violated the constraints. However, the design produced by the new framework defined in (42)–(46), with unity standard deviation scaling, a = 1, resulted in only 11.4% of the systems violating the constraints. Slightly increasing the standard deviation scaling to a = 1.25 resulting in only 3.5% of the systems failing to satisfy the constraints (see Figure 10). As described in Section 5.1.2, designers may choose to approach the problem with the constraints defined with *supremum* and/or *infimum* measures to guarantee the entire family of systems in the probability space will satisfy the constraints. However, this approach is relatively computationally costly. Therefore, a scaled standard deviation approach, as shown here, can yield an acceptably robust design. Similar design results are observed with regards to an active *holding* constraint as well, but are not shown for brevity.



Figure 9—Monte Carlo results (1000 runs) showing 59.6% of the systems in the probability space violate the *rattle* constraints when the deterministic optimal design is applied to an uncertain system; where $\underline{J}_{rattle} = \overline{J}_{rattle} = 0.152 [m]$ and $\overline{J}_{holding} = 0.034 [m]$.



Figure 10—Constraint violations from designs produced with the new framework can be controlled, or tuned, with the proper selection of the standard deviation scaling. Slightly increasing the scaling from a = 1 to a = 1.25 reduces the number of systems violating the constraints from 11.4% to 3.5%; where 1000 Monte Carlo simulations were used to determined the results; and $J_{rattle} = \overline{J}_{rattle} = 0.152 [m]$ and $\overline{J}_{holding} = 0.034 [m]$.

6 CONCLUSIONS

This work has presented a new framework for the parametric optimal design of uncertain ordinary differential equation systems. The framework allows practitioners to model sources of uncertainty using the Generalized Polynomial Chaos methodology and to solve the dynamics using a least-squares collocation method. Subsequently, statistical information from the uncertain dynamics can be included in formulations of the objective function and of the constraints, to perform optimal designs under uncertainty. Robust designs benefit from the new framework when the system is nonlinear, has active constraints, or is a multi-objective optimization problem. In the case of a multi-objective optimization problem, a constraint-based formulation of the problem was shown to produce an off-set Pareto optimal trade-off curve confirming the need to directly treat uncertainties during the optimal design phase. An optimal nonlinear vehicle suspension design problem, subject to parametric uncertainty, was used to illustrate how the new framework produces an optimal design that accounts for the entire family of systems within the associated probability space. This adds robustness to the design of the optimally performing system.

In future work the authors will expand the new framework to investigate optimal apportionment of uncertainty related to engineered components within a system such as to minimize cost while preserving the performance specifications of the overall system.

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